



# Potential radio frequency interference with the GPS L5 band for radio occultation measurements

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**Abstract.** Future radio occultation (RO) receivers are planned to utilize the newly implemented Global Positioning System (GPS) L5 band centered at 1176.45 MHz. Since there are currently no operational GPS L5 receivers used for space-based RO applications, the interference environment is unclear. Distance measuring equipment (DME) and tactical air navigation (TACAN) stations share the same frequency band as GPS L5. The signals from these stations have been identified as possible sources of interference for any GPS L5 receiver, including those used in RO applications. This study utilizes Systems Tools Kit (STK) simulations to gain insight into the power received by a RO satellite in low Earth orbit (LEO) from a DME–TACAN transmission as well as the amount of interfering stations. In order to confirm the validity of utilizing STK for communication purposes, a theoretical scenario was recreated as a simulation and the results were confirmed. Once the method was validated, STK was used to output a received power level aboard a RO satellite from a DME–TACAN station as well as a tool to detail the number of interfering DME–TACAN stations witnessed by a space-based RO receiver over time. The results indicated a large number of DME–TACAN stations transmitting at similar orientations as a receiving RO satellite, thereby leading to the possibility of signal degradation in an unclear interference environment.

## 1 Introduction

The Global Positioning System (GPS) L5 band centered at 1176.45 MHz is now being transmitted with the latest IIF satellite design (Van Dierendonck et al., 2000). This signal is part of the GPS modernization effort and offers civil users additional power, a higher chipping rate, and an updated signal modulation structure. As such, it is a promising signal transmission for scientific applications of GPS.

However, certain aeronautical navigation systems already occupy this frequency range. Distance measuring equipment (DME) and tactical air navigation (TACAN) systems offer potential sources of interference due to coexistence within the L5 band (Kim and Grabowski, 2003). These systems are comprised of an airborne interrogator and a ground-based transponder. A TACAN system is essentially a higher powered DME station used for military purposes. Due to the limited placements available within the Aeronautical Radio Navigation Services (ARNA) radio band for aviation use, the GPS L5 signal was placed within the already existing DME–TACAN band. The premise was that an aircraft using the system would only encounter a limited number of pulsed interfering signals, thereby allowing the interoperability between a GPS L5 receiver and a DME–TACAN signal. However, due to the higher number of interfering stations seen by a GPS radio occultation (RO) satellite in low Earth Orbit (LEO), the possibility for signal degradation for RO applications exists (Kim and Grabowski, 2003).

Interference incurred due to the coexistence of these systems degrades the carrier-to-noise ratio ( $C/N_0$ ) of a GPS L5 receiver. However, the compatibility of these systems is sufficient for most applications. The low power of a received GPS

L5 signal for terrestrial users on Earth from the GPS satellites has little if any impact for DME–TACAN operators. Furthermore, the pulsed localized nature of the DME–TACAN signals has minimal impact on terrestrial GPS L5 users as there are limited DME–TACAN sources in close proximity to any terrestrial user and code division multiple access (CDMA) modulation of GPS is robust against pulsed interference.

While most users of GPS L5 will experience minimal degradation from DME–TACAN interference, GPS RO is one such application in which even a slight degraded C/No would have a significant impact on results. This system is implemented today for use in weather forecasting and has been proven to be a very powerful and reliable tool. The architecture of a GPS RO system consists of a satellite in LEO receiving a signal from a GPS satellite. The LEO satellite houses a set of antennas pointed towards the limb of the Earth in order to detect and measure refraction as the signal propagates through the Earth's atmosphere. As a result of this directive orientation of the receiving antenna, these satellites may incur DME–TACAN interference that could obstruct RO data collection. The architecture of GPS RO will be discussed in further detail in the following section.

## 2 Background

### 2.1 Radio occultation

The utilization of GPS RO in weather forecasting has spurred a further advancement in forecasting accuracy. Utilizing the GPS satellite network, RO techniques leverage the stability and global coverage of the GPS network in order to provide higher-accuracy temperature, pressure, and humidity data (Healy et al., 2005). The process involves a sounding technique where a satellite emits a radio wave whose path is then perturbed by an intervening planetary atmosphere before reaching the receiver (Kursinski et al., 1997). Earth-based RO specifically involves a GPS satellite transmitting a signal to a receiving satellite orbiting in LEO. After the transmitted radio wave is refracted, phase and amplitude variation at the receiver is observed over time in order to define the refractive properties of the surrounding atmosphere (Melbourne, 2004). The refraction of the signal causes an excess phase in the dual-frequency carrier phase results as seen by the GPS receiver in LEO (Ware et al., 1996). By observing the degree of refraction, one can gain insight into the vertical distribution of atmospheric pressure, temperature, and humidity. The atmospheric depth of RO retrievals is currently limited by the available signal-to-noise Ratio (SNR). Additional SNR and increased signal-to-interference-plus-noise ratio (SINR) would allow for lower atmospheric data to be obtained.

Previous Earth-based occultation missions, such as GPS–Met and CHAMP (CHALLENGING Minisatellite Payload), improved upon numerical weather prediction (NWP)

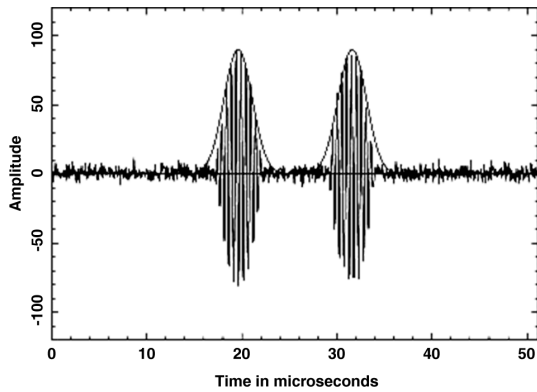
models when compared against the industry standard (Healy et al., 2005). All previous missions have utilized the L1 and L2 GPS frequencies and have exceeded expectations with respect to weather forecasting (Melbourne, 2004). However, the planned implementation of the L5 frequency in a dual-frequency configuration offers an opportunity to improve upon these results. The Formosat-7/COSMIC-2 (Constellation-Observing System for Meteorology, Ionosphere, and Climate) mission is a future joint mission between Taiwan and the United States that aims to utilize L5 receivers for reasons of increased power, overall improvement of signal structure, and the civil designation of the transmission (Mannucci et al., 2012). The TriG (Tri-GPS–GNSS–RO) receiver is one such receiver developed to utilize these characteristics of the L5 band (Esterhuizen et al., 2009).

### 2.2 Distance measuring equipment

The architecture of the DME system offers a method to determine distance from an aircraft to a ground station (Fisher, 2004). The DME architecture is comprised of an airborne interrogator and a ground-based transponder that operates in four codes ( $X$ ,  $Y$ ,  $W$ ,  $Z$ ). However, the  $X$  code is the only possible interferer with respect to the L5 frequency. The aircraft interrogates within a frequency range of 1025–1150 MHz, whereas the ground station transmits over frequencies between 1151 and 1213 MHz within  $X$  mode (Bastide et al., 2004). Therefore, any airborne interrogation within this architecture does not directly impinge upon any signal transmitted over the L5 frequency. A number of DME–TACAN ground stations, however, transmit within this frequency range and could become a source of interference for L5 transmissions. For this reason, DME–TACAN ground stations will be the focus for determining interoperability within the L5 frequency for GPS RO applications.

A DME ground station transmits in pulse pairs with a pulse period of 12  $\mu$ s and a half-amplitude pulse width of 3.5  $\mu$ s (Ostermeier, 2010). This signal structure can be seen in Fig. 1. In addition, DME stations either operate at a high power of 1000 W or at a low power of 100 W. During peak activity, a DME station transmits up to 2700 pulse pairs per second. The effective width of each pulse is defined to be 8  $\mu$ s taking into account a 1  $\mu$ s desaturation time for the receiver. Using this effective pulse width and the pulse pair rate previously defined, a single DME pulse duty cycle is calculated to be 0.0432 s<sup>-1</sup> (Roturier, 2001). Therefore, a single DME transmitter at its peak is seen 4.32 % of the time by an L5 receiver.

A TACAN station has many of the same characteristics as a DME station. However, unlike DME stations which transmit at a constant power of 100 or 1000 W, a TACAN station's transmission power ranges cyclically (sinusoidally at 135 and 15 Hz) up to 3500 W. These stations are consequently high-powered military versions of their DME counterpart.



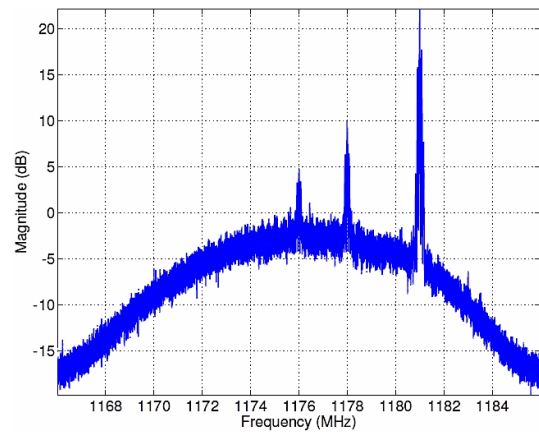
**Figure 1.** DME pulse pair measured at the Green Bank Telescope in West Virginia (Fisher, 2004).

A standard ground-based DME antenna gain pattern is maximum at  $4^\circ$  in elevation above the horizon and is omnidirectional in the azimuth. This orientation slightly above the horizon directly aligns with the directional gain pattern of a GPS RO satellite. The peak gain is  $+9.5$  dBi and the 3 dB beamwidth is  $6^\circ$  in elevation with vertical linear polarization (dB Systems Inc., 2013). As the orbiting satellite scans the limb of the Earth, gathering atmospheric data, it is in the main lobe of the directional beam of the DME station for a short period of time.

The United States and western Europe have high concentrations of DME stations, possibly inhibiting a GPS L5 RO receiver in LEO from properly functioning as the difference here is that the satellite will be illuminated by multiple DME stations. In the United States alone there are approximately 203 DME or TACAN ground stations that transmit within  $\pm 10$  MHz of the L5 center frequency of 1176.45 MHz. In assessing the impact to GPS RO, this is likely a conservative approach due to the fact that some RO receivers have wider bandwidths than  $\pm 10$  MHz (Esterhuizen et al., 2009). A receiver with a wider bandwidth will encounter a higher number of interfering DME stations. This is troubling for GPS RO scenarios because it offers up the possibility of receiver saturation, a situation in which no valid atmospheric data can be retrieved (ITU, 1998). Furthermore, the directive orientation of the receiver antenna pattern aboard a RO satellite with respect to a DME station increases the received power level from a DME station, increasing the cause for concern.

### 3 Pikes Peak L5 data collection

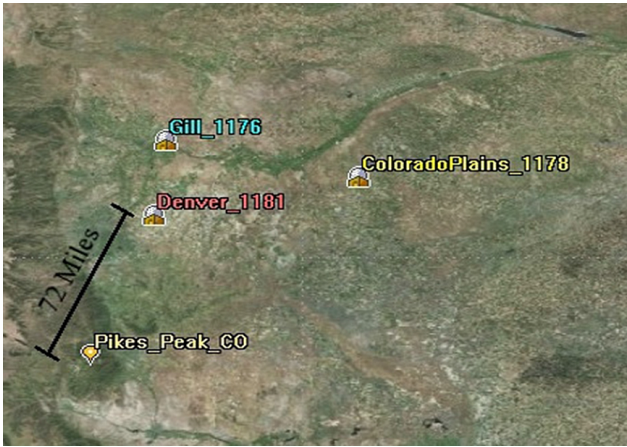
In order to assess the degree to which a directive antenna amplifies DME interference, it is useful to extract and analyze a real-world interference environment in which two separate antennas were compared. On 21 October 2011, data were collected on top of Pikes Peak mountain in Colorado at an elevation of approximately 4320 m by a team from the Uni-



**Figure 2.** Fourier transform of the time domain into the frequency domain as seen by a helical antenna atop Pikes Peak after post-processing. This measurement was taken on 21 October 2012 at 17:30 UTC.

versity of Colorado Boulder with the intention of conducting a ground-based RO measurement (Griggs, 2012). This test scenario was constructed in order to gain an understanding of the potential of the new GPS L5 signal for space-based RO. Collections of data from L1, L2, and L5 frequencies were gathered with two separate antennas. The first was a hemispherical survey grade antenna oriented vertically with approximately  $-3$  dBi gain toward the azimuth. The second was a helical antenna oriented horizontally and pointed  $38^\circ$  in the azimuth measured clockwise from north. The latter has a peak gain of  $+10$  dBi and a 3 dB beamwidth of  $45^\circ$ . The antenna used was the Q-par QHACP 1.2–1.6 GHz (Steatite Ltd., 2014). Since the data sheet is no longer available, a comparable helical antenna is the HE-0238-6. (R. A. Mayes Company Inc., 2014). Although this test cannot directly represent the results that a space-based receiver would yield, the data collected from this study have provided an insight into the amplification of DME signals when a high-gain antenna is employed. Further analysis seeking out the relative power values is currently ongoing.

Although the strength of the interference will undoubtedly be weaker in space, there will be a sharp rise in the number of DME stations affecting the GPS RO receiver. Within the L5 component of this collection, DME pulses can be seen within the data set. Figure 2 offers a depiction of the frequency domain as seen by the helical antenna pointed approximately northeast of Pikes Peak. Noting that the center frequency is 1176 MHz, DME stations transmitting at frequencies of 1176, 1178, and 1181 MHz can be seen within the collection. These frequencies correspond to DME stations in Gill, Colorado Plains, and Denver as shown in Fig. 3. On the other hand, DME interference is not directly observed in the data set collected by a survey grade antenna. The reason for this contrast is the differing orientations of the two antennas. The side-by-side comparison shown in Figs. 4 and 5 illuminates



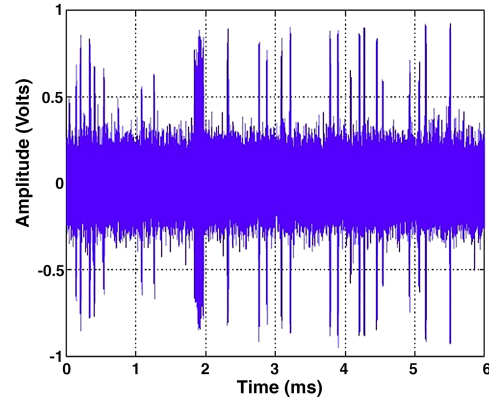
**Figure 3.** DME and TACAN stations encountered within Pikes Peak measurement as noted in Fig. 2. The helical antenna atop Pikes Peak was pointed  $38^\circ$  in the azimuth relative to north.

this stark contrast of interference between the data gathered by both antennas.

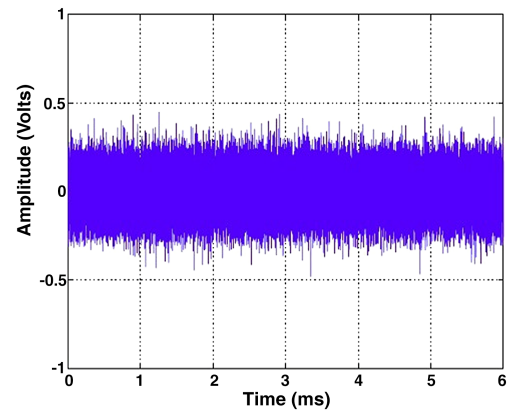
The interference experienced by the helical antenna registered considerably higher, greater than the thermal noise floor. The helical antenna is focused on the area of interest and therefore gathers visible DME interference. Similarly, a GPS L5 receiver used for RO applications utilizes a high-gain directive antenna whose gain is focused on the limb of the Earth (Wu et al., 2005). Further analysis was conducted in order to detail the interference that a GPS RO satellite may encounter due to DME pulses.

#### 4 Systems Tool Kit validation

Systems Tool Kit (STK) was utilized to attain a link budget for the received power of a DME station by a satellite in LEO. In order to establish the credibility of a space-based STK simulation, a scenario of a theoretical calculation was reconstructed within STK, and the results were compared to the theoretical solution. Roturier (2001) calculates the minimum pulse peak power at an aircraft's GPS receiving antenna under certain conditions. He defines the scenario as a receiving aircraft flying at an altitude of 12 192 m and a transmitting DME station located on the radio horizon from the aircraft's perspective. Both antennas were modeled as isotropic, and the DME radiated peak power,  $P_e$ , was set at 40 dBW. Inputting these specifications in Eq. (1) below yields an approximate minimum peak power received,  $P_1$ , of  $-107$  dBW, where  $G$  is the gain of the airborne GPS antenna set at 0 dB,  $\lambda$  represents the signal's wavelength equal to 25.5 cm, and  $d$  is the distance between transmitter and receiver equivalent to 246 NM.



**Figure 4.** Time domain as seen by helical antenna atop Pikes Peak after signal processing. Measurement taken on 21 October 2012 at 17:30 UTC.



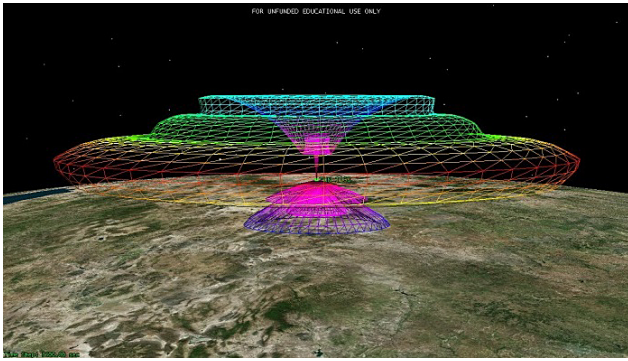
**Figure 5.** Time domain as seen by a standard survey grade antenna (hemispherical) atop Pikes Peak after signal processing. Measurement taken on 21 October 2012 at 17:30 UTC.

$$P_1 = P_e G \left( \frac{\lambda}{4\pi d} \right)^2 \quad (1)$$

Although Roturier (2001) simplifies this scenario by using isotropic antennas for both the DME stations and the GPS receiver, these identical parameters were recreated within STK and the results were compiled. The STK simulation outputted a value of  $-106.91$  dBW for the minimum pulse peak power received. The accuracy of this result when compared to the theoretical value supplies a level of integrity for using STK to compute a communication link budget.

#### 5 STK simulation and link budget results

In order to estimate the received power levels and range of a DME ground station as seen by a satellite in LEO, a simulation modeling these conditions was constructed within



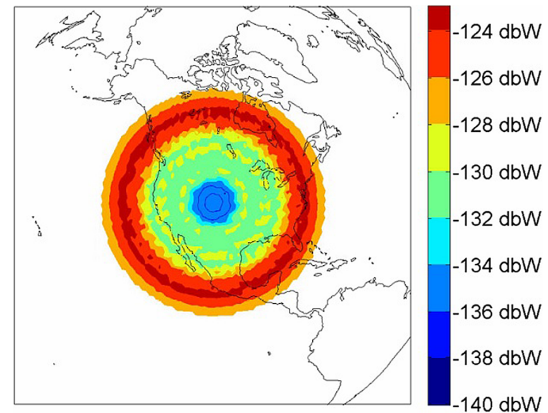
**Figure 6.** Systems Tool Kit model of dB Systems 5100A DME antenna pattern (Systems Tool Kit, 2014). Figure 10 provides a plot of the gain pattern with respect to elevation.

STK. A satellite database within STK was used to insert the Formosat-3 FM4 satellite into the simulation. This satellite is one of six in a constellation currently commissioned under the COSMIC-1 mission. The selection of this satellite therefore lends an accurate portrayal of a satellite that would house an L5 receiver for future RO missions. The gain pattern for the satellite's antenna was modeled in STK and the fore and aft elevation angles were set at  $27.38^\circ$  and  $27.16^\circ$ , scanning the limb of the Earth (Griggs, 2012). A single DME station was placed in Boulder, Colorado, for testing purposes, and a custom antenna pattern was modeled after the dB Systems Inc. 5100A high-performance DME antenna available on their website (dB Systems Inc., 2013). It should be noted that this model was chosen only as a representative pattern. Figure 6 is an image of the modeled gain pattern within STK.

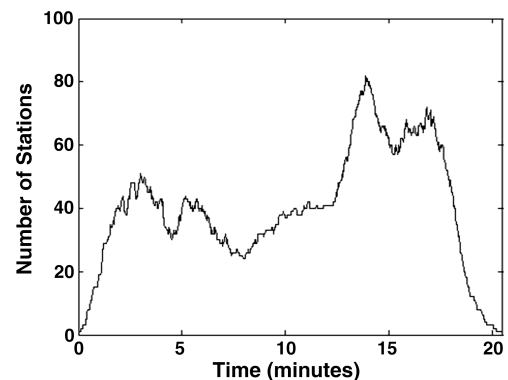
The transmission power of the DME station was set at 1000 W. This value was chosen based upon the fact that DME stations most commonly transmit at this high-power setting. It should be recalled that this model does not account for the low DME power setting of 100 W and the dynamic power ranges of a TACAN that can reach 3500 W. With the standard DME model in place, the simulation was progressed over a period of 6 months within STK, and a plot of the results is illustrated in Fig. 7.

The plot demonstrates a color-coded map of the received power with respect to the COSMIC satellite's position over the United States. This simulation indicates that, under ideal conditions, a DME transmission is received by the satellite at a maximum power level of  $-123$  dBW. This power level reaches a maximum when the satellite is within the main beam of the DME antenna. The received power then lessens until it abruptly ends as the satellite loses line of sight with the DME station.

In order to evaluate whether receiver saturation will be a potential problem for GPS RO satellites, an estimation of the time a single DME station interferes with a GPS L5 receiver and the total number of DME stations interfering at a given point in time are required. Due to the pulsed na-



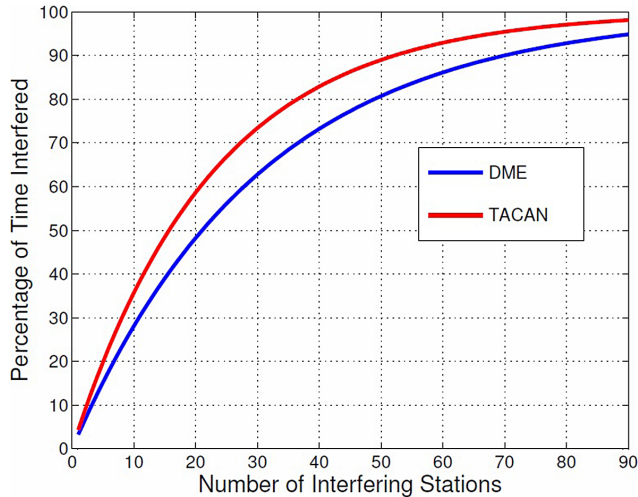
**Figure 7.** Formosat-3 FM4 satellite received power from a single DME station located at  $39.8125^\circ$  latitude and  $-104.661^\circ$  longitude. Results gathered from STK simulation.



**Figure 8.** Number of interfering DME stations with received power levels above  $-125$  dBW with respect to time as seen by the Formosat-3 FM4 satellite on 8 January 2013 from 16:02:21 to 16:22:50 UTC. The figure details the satellite's orbit across the United States from New Mexico to North Dakota and reaches a maximum when the satellite is orbiting over Manitoba, Canada. Results gathered from STK simulation.

ture of a DME signal, a GPS L5 receiver technically will not experience the DME transmission at all times. Recalling the calculated duty cycle of 4.3 % for a single DME station offers an estimate for the maximum time a DME station may transmit every second. In order to estimate how many DME stations would be interfering with a receiver in LEO at any given time, STK was utilized to provide the number of stations whose received power was greater than  $-125$  dBW, which was provided as an arbitrary constraint.

All 203 relevant DME stations were inserted into STK, and the number of interferers with respect to time was computed as the Formosat-3 FM4 satellite with the same antenna parameters as the previous STK simulation starting in the South Pacific Ocean traveled over the United States from New Mexico up towards North Dakota until all connections were lost in northern Canada. See Table 1 for the list of the



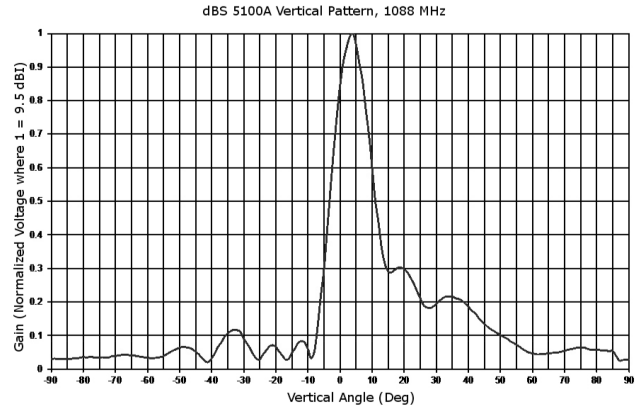
**Figure 9.** Percentage of time interfered as a function of the number of stations. This percentage is calculated through the use of Eq. (2).

relevant DME stations. The resulting plot is shown in Fig. 8. This figure indicates a maximum number of 82 stations transmitting with received powers of above  $-125$  dBW. This extreme case occurred when the satellite was over Manitoba, Canada. Due to the orientation of the DME gain pattern, it is consistent with the results that the highest concentration of signal is found on the edge of the network, not directly overhead. Due to the large amount of DME stations encountered by a space-based receiver, an overlap of the pulsed DME transmissions will occur. A separate calculation was implemented to provide insight into this overlap of the pulsed signals. The curves depicted in Fig. 9 illustrate the percentage of the time a receiver is being interfered with by DME and TACAN stations. The following equation defines this scenario as the number of interfering stations approaches 90, where  $I$  is the percent of time interfered,  $D$  is the duty cycle of a DME signal equivalent to 4.32 %, and  $N$  is the number of DME stations.

$$I = 1 - (1 - D)^N \quad (2)$$

## 6 Conclusions

A real-world validation of any simulation within STK is currently impossible due to the lack of space-based L5 GPS receivers with a representative RO pattern. With the future implementation of L5 receivers in space, the opportunity for experimental testing will be realized. However, without full knowledge of the parameters and conditions that a specific DME–TACAN station was operating under at the exact time of the data collection, any experiment will inherently be flawed. Uncertainties in specific antenna patterns, antenna efficiencies, receiver noise figure and other unknown variables



**Figure 10.** Plot of the gain pattern of 5100A DME antenna with respect to elevation angle (dB Systems Inc., 2013).

obscure any reasonable result. Although a space-based test cannot currently be conducted, other tests were undertaken in order to add to the credibility of the space-based STK simulation.

The single transmitting DME station and orbiting RO satellite simulation yielded a maximum received power level of  $-123$  dBW. A separate STK simulation including all of the relevant DME stations in the United States indicated that a GPS RO receiver in LEO may witness more than 80 stations at a time with received power levels above  $-125$  dBW. This number of relatively high-powered transmissions may cause substantial interference and possible saturation for a space-based GPS L5 receiver.

The resultant maximum received power should be noted as a conservative estimate due to restricting the study of the RO receiver bandwidth to within  $\pm 10$  MHz of the center frequency of 1176.45 MHz. Many RO receivers utilize a wide bandwidth with limited filtering in order to optimize data collection. However, this study implies that this approach may prove to have an adverse effect on the receivers' ability to gather data due to a greater number of interfering DME stations within the collected frequencies.

The resultant values may still not portray an entirely accurate estimate due to possible inaccuracies embedded in custom antenna patterns as well as the specific orientations for the antennas. A detailed analysis specifying correct transmission powers for each station could yield different results; however, using a 1000 W transmission power appears to be a suitable approach.

With these considerations in mind, the results gathered from the STK simulations indicate that L5 GPS receivers in LEO may experience interference caused by DME and TACAN stations, resulting in the possible saturation of the receivers. For the application of radio occultation, this interference could result in a loss of collected data as the satellite orbits over regions highly populated with DME stations, namely the United States and western Europe.

**Table 1.** Name, location, and corresponding transmission frequency of DME–TACAN stations within  $\pm 10$  MHz of 1176.45 MHz in the United States.

Name	Symbol	Freq (MHz)	Latitude	Longitude
ABILENE	ABI	1171	32.481328	−99.863456
ALBUQUERQUE	ABQ	1166	35.043797	−106.816314
AKRON	ACO	1178	41.107903	−81.201511
AKRON	AKO	1178	40.155578	−103.179739
ALAMOSA	ALS	1173	37.349158	−105.815536
ALMA	AMG	1185	31.536531	−82.508081
ARMEL	AML	1169	38.934597	−77.466694
NAPOLEON	ANX	1174	39.095425	−94.128836
WALNUT RIDGE	ARG	1179	36.109997	−90.953669
ASTORIA	AST	1174	46.161689	−123.880364
BALTIMORE	BAL	1185	39.171064	−76.661256
BRADFORD	BDF	1181	41.159731	−89.587872
BANGOR	BGR	1182	44.8418	−68.873964
BIG SPRING	BGS	1177	32.385589	−101.483683
BILLINGS	BIL	1179	45.808561	−108.624669
NASHVILLE	BNA	1175	36.136961	−86.684772
BOISE	BOI	1167	43.552811	−116.192131
BEAUMONT	BPT	1179	29.946056	−94.016222
BULLION	BQU	1179	40.759675	−115.761367
BROOKE	BRV	1179	38.336283	−77.352903
BIG SUR	BSR	1174	36.181294	−121.642114
BEATTY	BTY	1181	36.800583	−116.747647
BYERS	BVR	1169	39.765833	−103.928044
BOILER	BVT	1185	40.556119	−87.069319
BROADWAY	BWZ	1176	40.798433	−74.821833
BURLEY	BYI	1175	42.580244	−113.865853
BONHAM	BYP	1180	33.537486	−96.234094
COLUMBIA	CAE	1181	33.857247	−81.053906
CAMBRIDGE	CAM	1184	42.994289	−73.344019
CHADRON	CDR	1168	42.558772	−103.312147
CHICAGO HEIGHTS	CGT	1176	41.510006	−87.571544
CHARLESTON	CHS	1169	32.894319	−80.037814
CEDAR RAPIDS	CID	1175	41.887533	−91.785706
CHEROKEE	CKW	1184	41.755708	−107.581983
COLLEGE STATION	CLL	1167	30.605003	−96.420681
CHARLOTTE	CLT	1184	35.190472	−80.952
CEDAR CREEK	CQY	1182	32.185722	−96.218103
CRAIG	CRG	1179	30.338861	−81.509944
CUT BANK	CTB	1178	48.564944	−112.34325
CHESTER	CTR	1185	42.291319	−72.949394
DAVENPORT	CVA	1172	41.708542	−90.483306
COFIELD	CVI	1180	36.372914	−76.871544
COYLE	CYN	1168	39.817314	−74.431622
DAGGETT	DAG	1166	34.962458	−116.578167
DOLPHIN	DHP	1173	25.799964	−80.349036
DRYER	DJB	1170	41.358064	−82.161969
DAYTON	DQN	1179	40.016419	−84.396872
DUPONT	DQO	1174	39.678147	−75.607092
DRAKE	DRK	1175	34.702583	−112.48025
DOVE CREEK	DVC	1180	37.808739	−108.931272
MILE HIGH	DVV	1181	39.894694	−104.624333
DETROIT	DXO	1168	42.213136	−83.366664
PECK	ECK	1174	43.255886	−82.717931
NEEDLES	EED	1186	34.766003	−114.474103

Table 1. Continued.

Name	Symbol	Freq (MHz)	Latitude	Longitude
ELKINS	EKN	1176	38.914436	-80.099272
MEEKER	EKR	1186	40.067472	-107.924881
EL PASO	ELP	1186	31.815911	-106.281883
CENTRALIA	ENL	1184	38.420022	-89.159017
PEOTONE	EON	1166	41.269639	-87.791053
KEWANEE	EWA	1172	32.366808	-88.458369
NEW BERN	EWN	1170	35.073131	-77.045058
KEY WEST	EYW	1169	24.585878	-81.800475
FLAT ROCK	FAK	1167	37.528508	-77.828219
KALISPELL	FCA	1166	48.214103	-114.175892
FORT LAUDERDALE	FLL	1178	26.074739	-80.152472
FLORENCE	FLO	1186	34.232933	-79.657114
FORT DODGE	FOD	1169	42.611167	-94.294833
FORTUNA	FOT	1174	40.671272	-124.234539
WILLIAMSPORT	FQM	1178	41.338556	-76.774861
SIoux FALLS	FSD	1184	43.649531	-96.781164
GILLETTE	GCC	1180	44.347772	-105.543486
GARDEN CITY	GCK	1167	37.919053	-100.725064
GRAND FORKS	GFK	1177	47.954833	-97.185369
GOFFS	GFS	1178	35.131144	-115.176442
GLASGOW	GGW	1173	48.215269	-106.625461
GOODLAND	GLD	1185	39.387861	-101.692306
GILL	GLL	1176	40.503869	-104.553014
GOSHEN	GSH	1171	41.525186	-86.027972
GREAT FALLS	GTF	1185	47.449981	-111.412164
GUTHRIE	GTH	1179	33.778278	-100.336197
GALLUP	GUP	1185	35.476	-108.872611
GAVIOTA	GVO	1172	34.531308	-120.091083
BLUE MESA	HBU	1183	38.452153	-107.039792
HARTFORD	HFD	1183	41.641106	-72.547417
WHITEHALL	HIA	1171	45.861797	-112.169597
HILL CITY	HLC	1171	39.258747	-100.22585
HALLSVILLE	HLV	1176	39.113542	-92.128233
HOLSTON MOUNTAIN	HMV	1180	36.437056	-82.1296
HARVEY	HRV	1175	29.850194	-90.002983
HAMPTON	HTO	1170	40.919017	-72.316694
HUDSPETH	HUP	1184	31.568703	-105.376319
SCREAMING EAGLE	HXW	1183	36.675603	-87.495011
EL NIDO	HYP	1176	37.219431	-120.400217
HAYWARD	HYR	1168	46.019006	-91.4464
HAZEN	HZN	1175	39.516414	-118.997689
WICHITA	ICT	1172	37.745258	-97.583831
KINGFISHER	IFI	1181	35.805267	-98.003917
LOUISVILLE	IIU	1182	38.103464	-85.577436
WILLIAMS	ILA	1178	39.07115	-122.027244
WILDHORSE	ILR	1172	43.593122	-118.955044
KIRKSVILLE	IRK	1180	40.135022	-92.591714
COLLIERS	IRQ	1173	33.707353	-82.162064
WILL ROGERS	IRW	1175	35.358589	-97.609233
WILLIE	IWA	1167	33.303175	-111.651442
GLEN ROSE	JEN	1184	32.159589	-97.877681
JEFFERSON	JFN	1186	41.760122	-80.748106
JAMESTOWN	JHW	1181	42.188608	-79.121306
JULIAN	JLI	1174	33.140458	-116.585936
JAMESTOWN	JMS	1179	46.932872	-98.678769



Table 1. Continued.

Name	Symbol	Freq (MHz)	Latitude	Longitude
JANESVILLE	JVL	1177	42.557986	-89.105306
JUNIOR	JYU	1170	32.344611	-86.991269
LOS ANGELES	LAX	1170	33.933422	-118.432431
LAKE CHARLES	LCH	1168	30.141514	-93.105569
LUCIN	LCU	1170	41.362953	-113.840619
LINDEN	LDN	1177	38.854392	-78.205556
LEBANON	LEB	1171	43.679194	-72.215083
LEEVILLE	LEV	1169	29.175231	-90.104019
MARCONI	LFV	1181	42.017172	-70.037269
LINDEN	LIN	1182	38.074589	-121.003858
LITTLE ROCK	LIT	1173	34.677672	-92.180528
SALMON	LKT	1169	45.021311	-114.084236
MIDLAND	MAF	1182	32.009344	-102.190389
MARIANNA	MAI	1174	30.786222	-85.12445
MAPLES	MAP	1168	37.590767	-91.788569
MACON	MCN	1176	32.691222	-83.647181
MASON CITY	MCW	1183	43.094739	-93.329872
TRINITY	MHF	1170	29.546344	-94.747514
MANCHESTER	MHT	1178	42.868531	-71.369544
MUNCIE	MIE	1178	40.237294	-85.394036
MUSKEGON	MKG	1186	43.169242	-86.039383
MORMON MESA	MMM	1177	36.769278	-114.277472
MODESTO	MOD	1180	37.627375	-120.957867
MASSENA	MSS	1175	44.914444	-74.722664
MINA	MVA	1185	38.5653	-118.032853
MARTHAS VINEYARD	MVY	1179	41.396211	-70.612722
MOSES LAKE	MWH	1184	47.210864	-119.316817
MODENA	MXE	1166	39.918053	-75.670797
MOLINE	MZV	1178	41.321061	-90.638081
FALLON	NFL	1169	39.416864	-118.704869
TRUAX	NGP	1174	27.686278	-97.294742
BRUNSWICK	NHZ	1186	43.873514	-69.921911
LEMOORE	NLC	1167	36.344117	-119.966333
WHIDBEY ISLAND	NUW	1172	48.354936	-122.661786
YUMA	NYL	1171	32.6468	-114.613453
WOLBACH	OBH	1182	41.375736	-98.353594
OCALA	OCF	1171	29.177475	-82.226344
FOOTHILLS	ODF	1168	34.695872	-83.297661
ROGUE VALLEY	OED	1170	42.479575	-122.912933
KOKOMO	OKK	1169	40.527789	-86.058017
OKMULGEE	OKM	1183	35.693097	-95.865978
OLYMPIA	OLM	1168	46.971639	-122.901833
O'NEILL	ONL	1173	42.470503	-98.686922
CHICAGO O'HARE	ORD	1173	41.987672	-87.904886
WOODSIDE	OSI	1173	37.392672	-122.281828
VALDOSTA	OTK	1182	30.780444	-83.279728
NOTTINGHAM	OTT	1171	38.705867	-76.744744
PENDLETON	PDT	1181	45.698419	-118.938703
PIONEER	PER	1166	36.746531	-97.160156
PANAMA CITY	PFN	1177	30.216256	-85.680942
RICH MOUNTAIN	PGO	1169	34.680456	-94.609003
PEORIA	PIA	1186	40.680067	-89.7928
PALMDALE	PMD	1179	34.6314	-118.063822
PRINCETON	PNN	1177	45.329197	-67.704203
PASO ROBLES	PRB	1177	35.672469	-120.627111
PARIS	PRX	1170	33.542378	-95.448292

Table 1. Continued.

Name	Symbol	Freq (MHz)	Latitude	Longitude
PAWLING	PWL	1177	41.769772	-73.600553
POCKET CITY	PXV	1167	37.928319	-87.762381
POINT REYES	PYE	1171	38.079756	-122.867828
RAVINE	RAV	1180	40.553378	-76.599378
ROBBINSVILLE	RBV	1172	40.202389	-74.495014
RICHMOND	RIC	1175	37.50235	-77.320278
KREMMLING	RLG	1172	40.002642	-106.442489
WORLAND	RLY	1182	43.964139	-107.950833
REDWOOD FALLS	RWF	1167	44.467275	-95.128231
SAN MARCUS	RZS	1183	34.509528	-119.770992
SACRAMENTO	SAC	1186	38.443658	-121.551622
SOD HOUSE	SDO	1177	41.407056	-118.034722
SEA ISLE	SIE	1182	39.095503	-74.800333
SAN JOSE	SJC	1175	37.374711	-121.944667
SAN ANGELO	SJT	1185	31.374953	-100.454875
SLATE RUN	SLT	1173	41.512758	-77.970111
SIDON	SQS	1181	33.463861	-90.277333
STONEWALL	STV	1172	30.206758	-98.705756
SALEM	SVM	1177	42.408869	-83.594189
SQUAW VALLEY	SWR	1166	39.180322	-120.269614
SAYRE	SYO	1186	35.345161	-99.635347
TUBA CITY	TBC	1169	36.121325	-111.269586
TUCUMCARI	TCC	1170	35.182139	-103.598519
STANFIELD	TFD	1182	32.885856	-111.908733
ST THOMAS	THS	1184	39.933228	-77.950944
TWENTYNINE PALMS	TNP	1176	34.112236	-115.769908
TYRONE	TON	1183	40.735117	-78.331294
TULSA	TUL	1178	36.196261	-95.788108
TRAVERSE CITY	TVC	1180	44.667919	-85.549958
QUITMAN	UIM	1174	32.880403	-95.366753
QUINCY	UIN	1170	39.847875	-91.278925
TEXOMA	URH	1177	33.944186	-96.391836
CEDAR LAKE	VCN	1186	39.537672	-74.967144
VANDALIA	VLA	1177	39.093683	-89.162464
MOUNT VERNON	VNN	1172	38.361953	-88.807336
VULCAN	VUZ	1178	33.670186	-86.8998
FREEPORT	ZFP	1166	26.555256	-78.69785
ZUNI	ZUN	1168	34.965753	-109.154508

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